Research on Thermal Response Behavior of the Intumescent Coating at High Temperature: An Experimental and Numerical Study

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Abstract: Intumescent coating is able to provide effective fire protection with both practicality and aesthetics. In this study, expansion performance and thermal physical property experiments are firstly carried out to obtain the basic parameters at different temperatures. Then, the thermal response model of the concrete-filled steel tubular (CFST) structure under the protection of intumescent coating in a fire is established. Finally, based on the experimental data and thermal response model, the effects of initial thickness, expansion rate, intra-pore emissivity and reaction heat on the structure temperature are discussed in detail. The results of this study can provide guidance on intumescent coating formulation design, as well as fire protection design.

Keywords: intumescent coatings; thermal insulation performance; thermal response model; experimental study

1. Introduction

Concrete-filled steel tubular (CFST) structures are widely used in modern buildings due to their excellent mechanical and construction performance [1,2]. However, the mechanical performance of a CFST structure degrades significantly with increasing temperatures, and the structure cannot maintain its strength and stiffness at high temperatures [3]. Therefore, effective fire protection measures must be taken to reduce the temperature-rise rate of the CFST structure. Intumescent coating is able to provide efficient fire protection with relatively less thickness, and it is a common fireproofing material for CFST structures [4]. The intumescent coating will swell at high temperatures, and a porous carbon layer will be formed with relatively more thickness [5]. The porous carbon layer directs the path of external heat transfer into the protected structure, and the thermal conductivity of the porous carbon layer is extremely low [6]. Meanwhile, the physical and chemical changes of intumescent coatings will absorb a large amount of heat [7]. In summary, intumescent coating is able to reduce the temperature-rise rate of the protected structure with both practicality and aesthetics [8–10].

In the past decades, researchers conducted extensive studies on the thermal insulation performance of intumescent coatings. The studies can be divided into two categories. One type focused on the material design and formulation for intumescent coatings and developed new components and formulations to improve the physical and chemical performance of intumescent coatings at high temperatures. Vabdersall [11] put forward that the intumescent coating’s thermal performance was dependent on the quantity ratio of carbon, nitrogen and phosphorus. Reshentnikov [12–14] found that the best thermal insulation performance was achieved when the ratio of polyphosphate to polyol was 7:3. Studies showed that expandable graphite expanded at high temperatures and formed porous carbon layers with high porosity [15–18]. Researchers found that the proper addition of fillers
improved the intumescent coating’s thermal insulation performance. For instance, the use of magnesium hydroxide, titanium, zirconium, fire-resistant fibers and nanomaterials were able to improve the physical and chemical performance at high temperatures [18–25].

The other type of study focused on the thermal response behavior of intumescent coatings at high temperatures, including pyrolysis, expansion and heat-transfer mechanisms. This type of study provided theoretical guidance for the design and formulation of intumescent coatings. In the early years, semi-empirical models were developed to describe the expansion process [26–30]. Blasi and Colomba [31,32] established a mathematical model to characterize the pyrolysis behavior of the intumescent coating; the model described the expansion process with some empirical parameters. Some researchers [33–35] established comprehensive models by considering a combination of the pyrolysis, expansion and heat-transfer process. Li et al. [36,37] adopted the constant equivalent thermal resistance to indicate the thermal insulation performance of the intumescent coating. Cirpici et al. [38] used the bubble growth model to describe the expansion behavior of the intumescent coating. Zhang et al. [39] studied the couple thermal transfer mechanisms of the porous char layer, including thermal conduction, convection and radiation.

In this paper, expansion performance and thermal physical property experiments at high temperatures are carried out to obtain the thickness, thermal conductivity and density of intumescent coating. Then, a numerical model is established to describe the thermal response of CFST structures under the protection of intumescent coatings in a fire. Based on the experimental data and numerical model, this study examines the thermal response characteristics of the structure. The effect of initial thickness, expansion rate, intra-pore emissivity and reaction heat are discussed in detail. This results of this study can be applied to provide guidance on fire protection design.

2. Experiments

2.1. High-Temperature Expansion Performance Experiment

A high-temperature expansion performance experiment of a typical intumescent coating was carried out to obtain the variation in thickness with temperature. The intumescent coating used in the experiment was produced by Wuli Coating Co. The substrate was a surface-treated Q235 steel plate with a size of 120 mm × 120 mm × 3 mm. The initial thickness of the intumescent coating was 3 (±0.1) mm. The specimen was placed on the experimental table for dry maintenance. The instrument used for the experiment was the SX2-10-13A high-temperature electric furnace produced by Chongqing Xinbang Electric Furnace Company. The electric furnace was composed of two parts: the furnace body and the temperature-control system, as shown in Figure 1. The maximum range of the furnace is 1573 K, and a variety of temperature-rise rates can be set to meet the required test conditions. The main processes were as follows.

(a) The specimen is placed on the refractory brick and surrounded by a certain amount of sponge to prevent heat dissipation, as shown in Figure 2.
(b) The temperature of the furnace is set according to the working conditions and maintained for 1800 s to ensure full expansion at this temperature.
(c) After cooling, the expanded specimen is taken out and the thickness is measured by vernier caliper.

The expansion thickness and the expansion rate, which is defined as the ratio of the expansion thickness to the initial thickness, are shown in Table 1. In order to display the variation trend with the change in temperature more clearly, the data in the table are plotted as curves, as shown in Figure 3. It can be seen that in the initial stage (293 K to 573 K), the expansion thickness and rate remained unchanged with the increase in temperature; this is because the components of the intumescent coating have not yet decomposed. When the temperature reached 573 K, the expansion thickness and rate increased with the increase in temperature, which indicates that the decomposition reaction of the foaming agent and other components started to take place. Then, the expansion thickness and rate continued...
to increase and reached the maximum value at a certain temperature. Finally, the expansion thickness and rate started to decrease; this is due to the intumescent coating's oxidation at high temperatures.

Figure 1. Electric furnace.

Figure 2. Specimen in the furnace.
Table 1. Thickness and expansion rate (d3-373 K indicates the initial thickness is 3 mm and the experiment temperature is 373 K).

<table>
<thead>
<tr>
<th>Specimen</th>
<th>Initial Thickness</th>
<th>Test Thickness</th>
<th>Expansion Rate</th>
</tr>
</thead>
<tbody>
<tr>
<td>d3-373 K</td>
<td>3.13 mm</td>
<td>3.13 mm</td>
<td>1</td>
</tr>
<tr>
<td>d3-473 K</td>
<td>3.06 mm</td>
<td>3.06 mm</td>
<td>1</td>
</tr>
<tr>
<td>d3-573 K</td>
<td>3.10 mm</td>
<td>3.10 mm</td>
<td>1</td>
</tr>
<tr>
<td>d3-673 K</td>
<td>3.02 mm</td>
<td>9.57 mm</td>
<td>3.16</td>
</tr>
<tr>
<td>d3-773 K</td>
<td>3.17 mm</td>
<td>16.95 mm</td>
<td>5.35</td>
</tr>
<tr>
<td>d3-873 K</td>
<td>3.05 mm</td>
<td>21.60 mm</td>
<td>7.08</td>
</tr>
<tr>
<td>d3-973 K</td>
<td>3.07 mm</td>
<td>21.90 mm</td>
<td>7.13</td>
</tr>
<tr>
<td>d3-1073 K</td>
<td>3.11 mm</td>
<td>17.77 mm</td>
<td>5.71</td>
</tr>
</tbody>
</table>

Figure 3. Thickness and expansion rate variation with temperature.

2.2. Thermal Conductivity Experiment

The instrument used for the thermal conductivity experiment was the C-Therm thermal conductivity meter, as shown in Figure 4, which is based on the principle of “Modified Transient Plane Source”. The thermal conductivity of solids, liquids, powders and colloids could be determined quickly and accurately. The thermal conductivity could be measured from 0 to 500 W/(m⋅K), and the temperatures were from 273 K to 423 K.

Figure 4. C-Therm conductivity meter.
According to the results of the high-temperature expansion performance experiment, the state of the intumescent coating at different temperatures varied greatly, as shown in Figure 5. Therefore, the experiment of thermal conductivity was divided into a low-temperature stage and a high-temperature stage. In the low-temperature stage (273 K–573 K), the intumescent coating was in a dense block state, and we tested the thermal conductivity directly. However, in the high-temperature stage (673 K–1073 K), the intumescent coating reached the expansion temperature and was in a porous carbon layer state. We first pressed the porous carbon layer into powder, then tested the powder coating’s thermal conductivity. The specimens are shown in Figure 6.

Figure 5. State of the intumescent coating at different temperatures.

Figure 6. Specimen for the thermal conductivity experiment. (a) Block state; (b) powder state.

Figure 7 displays the thermal conductivities of the intumescent coating before expansion under different temperatures. It can be seen that the thermal conductivity increased with the increasing temperature. The equation for the thermal conductivity and the temperature was obtained by linear fitting, as Equation (1). The thermal conductivity at different temperatures (273–573 K) can be calculated by the following equation.

\[ \lambda_s = 0.00049T - 0.0405 \] (1)
Figure 6. Specimen for the thermal conductivity experiment. (a) Block state; (b) powder state.

Figure 7 displays the thermal conductivities of the intumescent coating before expansion under different temperatures. It can be seen that the thermal conductivity increased with the increasing temperature. The equation for the thermal conductivity and the temperature was obtained by linear fitting, as Equation (1). The thermal conductivity at different temperatures (273–573 K) can be calculated by the following equation:

$$\lambda_s = 0.000093T + 0.0337$$ (2)

$$\lambda_s = 0.000130T + 0.0270$$ (3)

$$\lambda_s = 0.000127T + 0.0324$$ (4)

$$\lambda_s = 0.000127T + 0.0362$$ (5)

$$\lambda_s = 0.000103T + 0.0475$$ (6)

Figure 8 displays the thermal conductivities of the intumescent coating after expansion under different temperatures. It can be seen that the thermal conductivity increased with the increasing temperature. By linear fitting, multiple equations for the thermal conductivity with temperature were obtained, as Equations (2)–(6). The thermal conductivity of the powder coating at different temperatures (673–1073 K) can be calculated by the following equations:

Figure 8. Variations of the thermal conductivity after expansion with temperature.

2.3. Density and Porosity Experiment

The density experiment of the intumescent coating was similar to the thermal conductivity experiment. In the low-temperature stage (273–573 K), a direct method was adopted for testing. In the high-temperature stage (673–1073 K), there were many pores in the carbon layer, the density was characterized into true density and apparent density and an indirect method was adopted for testing. The specimen is shown in Figure 9. The true density denotes the expanded carbon density without the pores, and the apparent density means the expanded carbon density within the pores. Lastly, the porosity can be calculated by Equation (7), which stands for the relationship of the true density to the apparent density:

$$\frac{1}{\rho} = 1 - \phi$$ (7)

where \(\rho\) and \(\rho_1\) denote the apparent density and true density in g/cm³, and \(\phi\) is the porosity.

Figure 7. Variations of the thermal conductivity before expansion with temperature.
2.3. Density and Porosity Experiment

The density experiment of the intumescent coating was similar to the thermal conductivity experiment. In the low-temperature stage (273–573 K), a direct method was adopted for testing. In the high-temperature stage (673–1073 K), there were many pores in the carbon layer, the density was characterized into true density and apparent density and an indirect method was adopted for testing. The specimen is shown in Figure 9. The true density denotes the expanded carbon density without the pores, and the apparent density means the expanded carbon density within the pores. Lastly, the porosity can be calculated by Equation (7), which stands for the relationship of the true density to the apparent density:

$$\phi = 1 - \frac{\rho}{\rho_1}$$  \hspace{1cm} (7)

where $\rho$ and $\rho_1$ denote the apparent density and true density in g/cm$^3$, and $\phi$ is the porosity.

![Figure 9. Specimen in the high-temperature stage.](image)

The measured density and the calculated porosity are shown in Table 2. It can be seen that in the low-temperature stage (373–573 K), the true density decreased from 0.791 g/cm$^3$ to 0.723 g/cm$^3$. In this stage, the water and volatility in the intumescent coating escaped and the mass of the coating decreased, whereas the volume remained unchanged. In the high-temperature stage (673–1073 K), the apparent density decreased from 0.297 g/cm$^3$ to 0.074 g/cm$^3$, and the true density decreased from 0.611 g/cm$^3$ to 0.240 g/cm$^3$. The coating started to expand, the mass decreased, and the volume continued to increase. The apparent density and true density of the expanded coating both decreased with the increasing temperature. In addition, the porosity of the intumescent coating increased from 0.514 to 0.722 in the temperature range (673–973 K), whereas it decreased from 0.722 to 0.692 in the temperature range (973–1073 K). The is due to the intumescent coating’s oxidation at high temperatures.
Table 2. Measured density and calculated porosity.

<table>
<thead>
<tr>
<th>$T$ (K)</th>
<th>Apparent Density (g/cm$^3$)</th>
<th>True Density (g/cm$^3$)</th>
<th>Porosity</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>1st</td>
<td>2nd</td>
<td>3rd</td>
</tr>
<tr>
<td>373</td>
<td>0.296</td>
<td>0.298</td>
<td>0.296</td>
</tr>
<tr>
<td>473</td>
<td>0.149</td>
<td>0.150</td>
<td>0.153</td>
</tr>
<tr>
<td>573</td>
<td>0.081</td>
<td>0.084</td>
<td>0.082</td>
</tr>
<tr>
<td>673</td>
<td>0.070</td>
<td>0.071</td>
<td>0.071</td>
</tr>
<tr>
<td>773</td>
<td>0.075</td>
<td>0.074</td>
<td>0.074</td>
</tr>
</tbody>
</table>

3. Numerical Simulation

3.1. Physical Model

A typical CFST structure was selected as the research object. According to GB 50936-2014 [40], the outer diameter of the CFST structure should not be less than 168 mm and the wall thickness should not be less than 3 mm. In this study, the outer diameter of the steel pipe was taken as 180 mm and the wall thickness was 3 mm. Figure 10 shows the physical model of the CFST structure under the protection of the intumescent coating. According to GB 14907-2018 [41], the thickness of the intumescent coating should not be less than 1.5 mm.

![Physical model of the CFST structure under the protection of intumescent coating.](image)

3.2. Governing Equation

The thermal response of the CFST structure under the protection of the intumescent coating in the fire is a transient heat-transfer process, and the numerical simulation follows the three-dimensional unsteady heat conduction differential equation:

$$
\frac{\partial}{\partial t} (Tc\rho) = \frac{\partial}{\partial x} \left( \lambda \frac{\partial T}{\partial x} \right) + \frac{\partial}{\partial y} \left( \lambda \frac{\partial T}{\partial y} \right) + \frac{\partial}{\partial z} \left( \lambda \frac{\partial T}{\partial z} \right) \tag{8}
$$

where $T$ means temperature in K, $c$ is the thermal capacity in J/(kg·K), $\rho$ is the true density from the density experiment in kg/m$^3$, $x$, $y$, $z$ denote the coordinate system in m and $t$ represents the time in s.
The temperature change along the axial direction of the CFST structure was much smaller than that along the radial direction of the structure, so the temperature change along the axial direction of the structure can be ignored:

$$
\frac{\partial}{\partial z} \left( \lambda \frac{\partial T}{\partial z} \right) = 0
$$  (9)

Substituting Equation (9) into (8), the governing equation for the thermal response of the CFST structure under the protection of the intumescent coating is obtained as follows:

$$
\frac{\partial}{\partial t} \left( Tc \rho \right) = \frac{\partial}{\partial x} \left( \lambda \frac{\partial T}{\partial x} \right) + \frac{\partial}{\partial y} \left( \lambda \frac{\partial T}{\partial y} \right)
$$  (10)

The solution to Equation (10) requires the thermal property parameters of the intumescent coating and the CFST structure. The thermal property parameters of the intumescent coating in this study were obtained through the experiments, and the thermal property parameters of the CFST were taken based on GB 50936-2014 [40].

3.3. Initial and Boundary Conditions

The initial condition is as follows:

$$
T(x,y)|_{t=0} = T_0
$$  (11)

where $T_0$ denotes the ambient temperature with a value of 293 K.

Solving the above heat-transfer governing equation also requires boundary conditions. When the CFST structure is under fire, it is usually understood that external heat is transferred to the structure by means of thermal convection and radiation, as shown in Figure 11. The boundary conditions can be expressed as:

$$
q = q_{\text{conv}} + q_{\text{rad}}
$$  (12)

where $q$, $q_{\text{conv}}$ and $q_{\text{rad}}$ represent, respectively, the total heat flux, convective heat flux and radiative heat flux.

![Figure 11. Boundary conditions of the numerical model.](image)

The convective heat flux is defined as:

$$
q_{\text{conv}} = h_f \left( T_f - T_s \right)
$$  (13)
where \( h_f \) is the convective heat-transfer coefficient in W/(m\(^2\)-K), the value of \( h_f \) is in the range from 10 W/(m\(^2\)-K) to 20 W/(m\(^2\)-K) and 15 W/(m\(^2\)-K) is adopted in the numerical simulation according to Gillet [42]. \( T_f \) is the gas temperature, and \( T_s \) is the surface temperature of the intumescent coating.

The radiative heat flux is defined as [43]:

\[
q_{rad} = \varepsilon_{sys} \sigma (T_f^4 - T_s^4)
\]

where \( \varepsilon_{sys} \) denotes the system emissivity with a value of 0.75, and \( \sigma \) is the Boltzmann constant with a value of 5.67 \times 10^{-8} \text{ W/(m}^2\text{-K})^4\).

According to the boundary conditions, the gas temperature \( T_f \) is the main external factor affecting the heat transfer into the structure. Therefore, it is important to describe the temperature-rise history of the gas temperature accurately. Various fire scenarios have some differences in the temperature-rise models. The fire scenarios generally have two forms: (1) fire dominated by hydrocarbon substances, which is usually caused by petroleum; (2) fire dominated by fibrous substances, which is usually caused by wood, paper and textiles.

For the fibrous substances fire, the ISO 834 model is a commonly used model and can be expressed as:

\[
T_f = T_0 + 345 \log \left( \frac{t}{75} + 1 \right)
\]

Figure 12 shows the gas temperature curve used in the simulations; it can be seen from the figure that the gas temperature increased with time.

![Figure 12. Gas temperature curve used in the simulations.](image)

3.4. Treatments of the Nonlinear Problem in the Numerical Simulation

3.4.1. Heat-Absorption Effect in the Pyrolysis

The intumescent coating pyrolysis will absorb part of the external heat and thus reduces the temperature-rise rate of the protected structure, so the heat-absorption effect should be considered in the numerical simulation. The pyrolysis is treated as a special phase-change process, and the equivalent heat capacity method is used to solve the nonlinear problem in the phase change.

The sensible heat capacity \( c_{p,e} \) is defined as:

\[
c_{p,e} = \begin{cases} 
\frac{c_{p,s} - c_{p,l}}{2} + \frac{L}{T_l - T_s} & T \leq T_s \\
\frac{c_{p,s} + c_{p,l}}{2} & T_s < T < T_l \\
c_{p,l} & T \geq T_l 
\end{cases}
\]

where \( c_{p,s} \) and \( c_{p,l} \) represent, respectively, the sensible heat capacity before and after phase change; the value is 1000 J/(kg-K) and 2000 J/(kg-K). \( L \) is the latent heat; the value is 100 kJ/kg. \( T_s \) and \( T_l \) denote, respectively, the temperature before and after phase change, with values of 573 K and 873 K.
### 3.4.2. Nonlinear Size Effect in the Expansion

In the high-temperature expansion performance experiment, the thickness at different temperatures was obtained. Although the thickness of the coating changed with temperature, a fixed region was still used for the numerical simulation by considering the efficiency and feasibility. The change in the intumescent coating’s thickness is converted into the change in the thermal conductivity equivalently, which can simplify the numerical simulation and reduce the calculation cost. The equivalent conversion processes are as follows:

(a) Using Equation (1) to calculate the thermal conductivity before expansion and using Equations (2)–(6) to calculate the thermal conductivity after expansion, the results are shown in Table 3.

(b) An equivalent thermal conductivity model of the intumescent coatings is adopted to calculate the thermal conductivity [44]:

#### Table 3. Thermal conductivity of the intumescent coating.

<table>
<thead>
<tr>
<th>( T ) (K)</th>
<th>293</th>
<th>373</th>
<th>473</th>
<th>573</th>
<th>673</th>
<th>773</th>
<th>873</th>
<th>973</th>
<th>1073</th>
</tr>
</thead>
<tbody>
<tr>
<td>( \lambda_s ) (W/(m·K))</td>
<td>0.10</td>
<td>0.14</td>
<td>0.19</td>
<td>0.24</td>
<td>0.29</td>
<td>0.34</td>
<td>0.39</td>
<td>0.44</td>
<td>0.49</td>
</tr>
</tbody>
</table>

\[
\lambda_{eq} = \frac{\lambda_s \phi^\frac{2}{3} + (1 - \phi^\frac{2}{3}) \lambda_g}{\lambda_s (\phi^\frac{2}{3} - \phi) + (1 - \phi^\frac{2}{3}) + \phi} \lambda_s
\]

where \( \lambda_s \) denotes the intumescent coating’s thermal conductivity, \( \lambda_g \) denotes the air thermal conductivity, and \( \lambda_{eq} \) denotes the equivalent thermal conductivity. \( \phi \) indicates the porosity, \( d \) is the average diameter of the pore with a value of \( 3.495 \times 10^{-3} \) m, and \( \varepsilon \) is the intra-pore emissivity with a value of 0.8. The equivalent thermal conductivity of the intumescent coating at different temperatures is calculated by Equation (17). The results are shown in Table 4.

#### Table 4. Equivalent thermal conductivity of the intumescent coating.

<table>
<thead>
<tr>
<th>( T ) (K)</th>
<th>293</th>
<th>373</th>
<th>473</th>
<th>573</th>
<th>673</th>
<th>773</th>
<th>873</th>
<th>973</th>
<th>1073</th>
</tr>
</thead>
<tbody>
<tr>
<td>( \lambda_s ) (W/(m·K))</td>
<td>0.10</td>
<td>0.14</td>
<td>0.19</td>
<td>0.24</td>
<td>0.29</td>
<td>0.34</td>
<td>0.39</td>
<td>0.44</td>
<td>0.49</td>
</tr>
<tr>
<td>( \lambda_g ) (W/(m·K))</td>
<td>0.04</td>
<td>0.06</td>
<td>0.08</td>
<td>0.12</td>
<td>0.18</td>
<td>0.25</td>
<td>0.34</td>
<td>0.45</td>
<td>0.59</td>
</tr>
<tr>
<td>( \phi )</td>
<td>0.514</td>
<td>0.655</td>
<td>0.705</td>
<td>0.722</td>
<td>0.692</td>
<td>0.705</td>
<td>0.722</td>
<td>0.692</td>
<td>0.722</td>
</tr>
<tr>
<td>( \lambda_{eq} ) (W/(m·K))</td>
<td>0.10</td>
<td>0.14</td>
<td>0.19</td>
<td>0.24</td>
<td>0.23</td>
<td>0.28</td>
<td>0.35</td>
<td>0.44</td>
<td>0.56</td>
</tr>
</tbody>
</table>

(c) Assuming that the thickness of the intumescent coating remains unchanged, the thickness change is equivalently converted into the thermal conductivity change by Equation (18). As a result, the input values in the simulation calculation are obtained, which can be seen in Table 5:

#### Table 5. The input value in the numerical simulation.

<table>
<thead>
<tr>
<th>( T ) (K)</th>
<th>293</th>
<th>373</th>
<th>437</th>
<th>573</th>
<th>673</th>
<th>773</th>
<th>873</th>
<th>973</th>
<th>1073</th>
</tr>
</thead>
<tbody>
<tr>
<td>( n )</td>
<td>1</td>
<td>1</td>
<td>1</td>
<td>1</td>
<td>3.16</td>
<td>5.35</td>
<td>7.08</td>
<td>7.13</td>
<td>5.71</td>
</tr>
<tr>
<td>( \lambda_{eq} ) (W/(m·K))</td>
<td>0.10</td>
<td>0.14</td>
<td>0.19</td>
<td>0.24</td>
<td>0.07</td>
<td>0.05</td>
<td>0.05</td>
<td>0.06</td>
<td>0.10</td>
</tr>
</tbody>
</table>

\[
\lambda_{eq}' = \frac{\lambda_{eq}}{n}
\]

where \( \lambda_{eq}' \) represents the input value in the numerical simulation, \( n \) denotes the expansion rate from the expansion experiment, and the value is displayed in Table 1.
3.5. Results and Discussion

3.5.1. Finite Element Model and Feature Point

The governing equation with initial and boundary conditions was solved numerically using the commercial software Comsol Multiphysics (Version 5.5, Comsol Inc., Stockholm, Sweden). The quarter-symmetric model was adopted for lower cost. The built-in grid module was used for grid generation, and the grid structure with a legend of quality is shown in Figure 13. The closer the value to 1, the better the quality of the grid. The total number of grids was 20,334 and the average value was 0.95. The time step in the calculation was 0.01 s. Figure 14 shows the location of the feature point 1, which was used for numerical analysis. In this section, numerical simulations are carried out to investigate the temperature variation of point 1 with time. The effects of initial thickness (0, 2 mm, 3 mm, 4 mm, 5 mm), expansion rate (0.1 n, 0.2 n, n, 5 n, 10 n), intra-pore emissivity (0, 0.2, 0.4, 0.6, 0.8, 1) and reaction heat (10 kJ/kg, 100 kJ/kg, 1000 kJ/kg, 5000 kJ/kg, 10,000 kJ/kg, 20,000 kJ/kg) are studied in detail.

Figure 13. Grid structure of the model.

Figure 14. Location of the feature point.
3.5.2. The Effect of Initial Thickness

Figure 15 shows the temperature variation of point 1 with different initial thicknesses. Tables 6 and 7 present, respectively, the temperature and the temperature-rise rate of point 1 with different initial thicknesses at different times. It can be seen that temperature of point 1 was the highest when the initial thickness was 0. This indicates that the intumescent coating has a thermal insulation effect on the protected structure and can thus reduce the temperature and temperature-rise rate of the protected structure. In addition, the greater the thickness, the lower the temperature of point 1. This means that the intumescent coating’s thermal insulation performance increases with the increasing initial thickness. Furthermore, the effect of the initial thickness on the thermal insulation performance is also related to the time interval, and it was found that the best effect occurred in the range from 2000–4000 s.

![Temperature variation over time with different initial thicknesses.](image)

Table 6. Temperature with different initial thicknesses at different times.

<table>
<thead>
<tr>
<th>Time (s)</th>
<th>0</th>
<th>2 mm</th>
<th>3 mm</th>
<th>4 mm</th>
<th>5 mm</th>
</tr>
</thead>
<tbody>
<tr>
<td>500</td>
<td>593.3</td>
<td>473.2</td>
<td>437.4</td>
<td>409.7</td>
<td>387.9</td>
</tr>
<tr>
<td>1000</td>
<td>793.1</td>
<td>569.1</td>
<td>524.3</td>
<td>488.7</td>
<td>460.3</td>
</tr>
<tr>
<td>2000</td>
<td>1019.0</td>
<td>660.5</td>
<td>608.3</td>
<td>573.5</td>
<td>543.9</td>
</tr>
<tr>
<td>4000</td>
<td>1181.7</td>
<td>840.9</td>
<td>744.0</td>
<td>684.8</td>
<td>643.7</td>
</tr>
<tr>
<td>10,000</td>
<td>1354.6</td>
<td>1173.5</td>
<td>1078.7</td>
<td>989.9</td>
<td>913.1</td>
</tr>
</tbody>
</table>

Table 7. Temperature-rise rate with different initial thicknesses at different times.

<table>
<thead>
<tr>
<th>Time Interval (s)</th>
<th>0</th>
<th>2 mm</th>
<th>3 mm</th>
<th>4 mm</th>
<th>5 mm</th>
</tr>
</thead>
<tbody>
<tr>
<td>0–500</td>
<td>0.60</td>
<td>0.36</td>
<td>0.29</td>
<td>0.23</td>
<td>0.19</td>
</tr>
<tr>
<td>500–1000</td>
<td>0.47</td>
<td>0.19</td>
<td>0.17</td>
<td>0.16</td>
<td>0.14</td>
</tr>
<tr>
<td>1000–2000</td>
<td>0.23</td>
<td>0.09</td>
<td>0.08</td>
<td>0.08</td>
<td>0.08</td>
</tr>
<tr>
<td>2000–4000</td>
<td>0.08</td>
<td>0.09</td>
<td>0.07</td>
<td>0.06</td>
<td>0.05</td>
</tr>
<tr>
<td>4000–10,000</td>
<td>0.03</td>
<td>0.06</td>
<td>0.06</td>
<td>0.05</td>
<td>0.04</td>
</tr>
</tbody>
</table>

3.5.3. The Effect of Expansion Rate

Figure 16 shows the temperature variation of point 1 with different expansion rates. Tables 8 and 9 present, respectively, the temperature and temperature-rise rate of point 1
with different expansion rates at different times. The expansion rate \( n \) was obtained from the expansion performance experiment. It can be seen that the greater the expansion rate, the lower the temperature of point 1. This indicates that the intumescent coating’s thermal insulation performance increased with the increasing expansion rate. This is because the increasing expansion rate led to a longer path for heat transfer. In addition, the curves are very close to each other when the expansion rates were \( 5n \) and \( 10n \). This means that thermal insulation performance no longer significantly improved when the expansion rate increased to a certain level. In addition, more expansion may lead to a more fragile char that may get blown off the surface compared with a thinner, more durable char. However, this paper focuses on the heat-transfer mechanism, and the numerical model does not include the effects of the durability of the char itself.

![Figure 16. Temperature variation over time with different expansion rates.](image)

**Table 8.** Temperature with different expansion rates at different times.

<table>
<thead>
<tr>
<th>Time (s)</th>
<th>Temperature of Point 1 (K)</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>0.1 ( n )</td>
</tr>
<tr>
<td>500</td>
<td>467.3</td>
</tr>
<tr>
<td>1000</td>
<td>682.0</td>
</tr>
<tr>
<td>2000</td>
<td>906.7</td>
</tr>
<tr>
<td>4000</td>
<td>1130.1</td>
</tr>
<tr>
<td>10,000</td>
<td>1330.9</td>
</tr>
</tbody>
</table>

**Table 9.** Temperature-rise rate with different expansion rates at different times.

<table>
<thead>
<tr>
<th>Time Interval (s)</th>
<th>Temperature-Rise Rate of Point 1 (K/s)</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>0.1 ( n )</td>
</tr>
<tr>
<td>0–500</td>
<td>0.35</td>
</tr>
<tr>
<td>500–1000</td>
<td>0.43</td>
</tr>
<tr>
<td>1000–2000</td>
<td>0.22</td>
</tr>
<tr>
<td>2000–4000</td>
<td>0.11</td>
</tr>
<tr>
<td>4000–10,000</td>
<td>0.03</td>
</tr>
</tbody>
</table>

3.5.4. The Effect of Intra-Pore Emissivity

Figure 17 shows the temperature variation of point 1 with different intra-pore emissivity. Tables 10 and 11 present, respectively, the temperature and temperature-rise rate of point 1 with different intra-pore emissivity at different times. The intra-pore emissivity of 0 means that radiation in the carbon layer was not considered. From the figure, the
greater the intra-pore emissivity, the higher the temperature of point 1. This indicates that the intumescent coating’s thermal insulation performance decreased with the increasing intra-pore emissivity. The reason for this is that the increase in intra-pore emissivity enhances the intro-pore thermal radiation, thereby leading to an increase in equivalent thermal conductivity. Comparing the temperature variation at an intra-pore emissivity of 0 and 1, the temperature difference reached 326.3 K at \( t = 10,000 \) s. This means that thermal radiation in the carbon layer formed by the intumescent coating at high temperatures should be considered.

### Table 10. Temperature with different intra-pore emissivity at different times.

<table>
<thead>
<tr>
<th>Time (s)</th>
<th>0</th>
<th>0.2</th>
<th>0.4</th>
<th>0.6</th>
<th>0.8</th>
<th>1</th>
</tr>
</thead>
<tbody>
<tr>
<td>1000</td>
<td>507.3</td>
<td>510.2</td>
<td>517.6</td>
<td>522.5</td>
<td>524.3</td>
<td>527.9</td>
</tr>
<tr>
<td>2000</td>
<td>571.5</td>
<td>578.6</td>
<td>592.7</td>
<td>602.1</td>
<td>608.3</td>
<td>610.6</td>
</tr>
<tr>
<td>4000</td>
<td>630.5</td>
<td>651.6</td>
<td>692.7</td>
<td>716.6</td>
<td>744.0</td>
<td>772.3</td>
</tr>
<tr>
<td>10,000</td>
<td>780.4</td>
<td>878.0</td>
<td>975.1</td>
<td>1020.6</td>
<td>1078.7</td>
<td>1106.7</td>
</tr>
</tbody>
</table>

### Table 11. Temperature-rise rate with different intra-pore emissivity at different times.

<table>
<thead>
<tr>
<th>Time Interval (s)</th>
<th>0</th>
<th>0.2</th>
<th>0.4</th>
<th>0.6</th>
<th>0.8</th>
<th>1</th>
</tr>
</thead>
<tbody>
<tr>
<td>0–1000</td>
<td>0.21</td>
<td>0.22</td>
<td>0.22</td>
<td>0.23</td>
<td>0.23</td>
<td>0.23</td>
</tr>
<tr>
<td>1000–2000</td>
<td>0.06</td>
<td>0.07</td>
<td>0.08</td>
<td>0.08</td>
<td>0.08</td>
<td>0.09</td>
</tr>
<tr>
<td>2000–4000</td>
<td>0.03</td>
<td>0.04</td>
<td>0.05</td>
<td>0.06</td>
<td>0.07</td>
<td>0.08</td>
</tr>
<tr>
<td>4000–10,000</td>
<td>0.02</td>
<td>0.04</td>
<td>0.05</td>
<td>0.05</td>
<td>0.06</td>
<td>0.06</td>
</tr>
</tbody>
</table>

### 3.5.5. The Effect of Reaction Heat

In the calculations, the latent heat \( L \) represents the reaction heat, but \( L \) does not appear in the thermal governing equation (Equation (8)). Therefore, the equivalent heat capacity method (Equation (16)) was used to transition. Figure 18 shows the temperature variation of point 1 with different reaction heats. Tables 12 and 13 present, respectively, the temperature and temperature-rise rate of point 1 with different reaction heats at different times. The reaction heat represents the intumescent coating heat-absorb effect due to physical and chemical changes. From the figure, the greater the reaction heat, the lower the temperature of point 1. This indicates that the intumescent coating’s thermal insulation performance increased with the increasing reaction heat. The reason for this is that the heat-absorb
effect led to a decrease in temperature rate. However, the effect of the reaction heat on the thermal insulation’s performance is limited. From the figure and tables, the maximum temperature difference is 68.9 K at \( t = 1151 \) s when the reaction heat increases from 10 kJ/kg to 10,000 kJ/kg.

![Figure 18. Temperature variation over time with different reaction heats.](image)

**Table 12.** Temperature with different reaction heats at different times.

<table>
<thead>
<tr>
<th>Time (s)</th>
<th>Temperature of Point 1 (K)</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>10 kJ/kg</td>
</tr>
<tr>
<td>1000</td>
<td>525.3</td>
</tr>
<tr>
<td>2000</td>
<td>610.2</td>
</tr>
<tr>
<td>10,000</td>
<td>1080.16</td>
</tr>
</tbody>
</table>

**Table 13.** Temperature-rise rate with different reaction heats at different times.

<table>
<thead>
<tr>
<th>Time Interval (s)</th>
<th>Temperature-Rise Rate of Point 1 (K/s)</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>10 kJ/kg</td>
</tr>
<tr>
<td>0–1000</td>
<td>0.23</td>
</tr>
<tr>
<td>1000–2000</td>
<td>0.08</td>
</tr>
<tr>
<td>2000–10,000</td>
<td>0.06</td>
</tr>
</tbody>
</table>

4. Conclusions

Expansion performance and thermal physical property experiments were carried out to obtain the basic data of intumescent coating at high temperatures. The thermal response model of the CFST structure under the protection of intumescent coating in a fire was established, and the effects of initial thickness, expansion rate, intra-pore emissivity and reaction heat on the structure’s temperature were discussed based on the numerical simulation. The following conclusions could be drawn:

1. The thickness of the intumescent coating stayed the same at low temperatures, then increased rapidly with the increasing temperature, finally decreasing due to oxidation at high temperatures. The thermal conductivity increased with the increasing temperature, and the calculation equations at different temperature were obtained.

2. The initial thickness had a positive effect on the intumescent coating’s thermal insulation performance. As the initial thickness increased, the thermal insulation performance improved.
The expansion rate had a remarkable effect on the intumescent coating’s thermal insulation performance, but the effect was limited when the expansion rate increased to a certain level. In the material formulation design, appropriately increasing the quantity of the foaming agent could effectively improve the thermal insulation performance.

The intra-pore emissivity had a negative effect on the intumescent coating’s thermal insulation performance. Thermal radiation in the porous carbon layer strengthened the coupled heat transfer. In the material-formulation design, specific fillers could be used to decrease the intra-pore emissivity, thereby improving the thermal insulation performance.

The reaction heat had a slightly positive effect on the intumescent coating’s thermal insulation performance. In the material formulation design, modifying the components could strengthen the heat-absorption effect, thus improving the thermal insulation’s performance.

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Conflicts of Interest: The authors declare no conflict of interest.

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