Abstract
This paper deals with electrical propulsion system for light airplane or paraglider. The electrical drive consists of electrical motor, frequency converter and the batterypack. Electrical motor used in a light paraglider was designed as a permanent magnet synchronous motor (PMSM) in the Research and Development Centre of Electrical Machines KOMEL, Poland. Parameters of the motor, frequency converter and battery pack were shown. In this article, an influence of different propellers (1 fixed, 2 adjustable) and results of laboratory test was described. For increasing the efficiency of the electric propulsion, influence of switching frequency of the converter were tested and presented. The results of laboratory test were discussed.

Keywords: Battery, permanent magnet synchronous motor, motor design, paraglider, electric drives

1 Introduction
Aviation waits for environment friendly drives and propulsion systems. This paper deals with electrical propulsion for a light airplane or paraglider which was shown in Figure 1.

The electrical drive consists of an electrical motor, frequency converter and batterypack. Electrical motor used in light paraglider was designed as a permanent magnet synchronous motor (PMSM) in Research and Development Centre of Electrical Machines KOMEL, Poland. The paraglider in flight is presented in Figure 2.
The designed propulsion system was mounted in paraglider’s trike. Light two person paraglider is able to fly for about 25 minutes with two persons and almost 1 hour with one person. The electric motor can be turned off during the flight what conserves the energy.

2 Electric motor

Parameters of the electric motor type SMwsK90L-8 are shown in Table 1. Electrical motor was designed as an interior permanent magnet synchronous motor (IPM), air cooled, 25 kW, frame size 90 mm. The rotor of the motor has a special slots for magnets. Cross section of electromagnetic circuit of electric motor is shown in Fig. 3. Electric motor mounted in paraglider with the propeller is presented in Fig. 4.

Table 1: Parameters of electric motor SMwsK90L-8

<table>
<thead>
<tr>
<th>Nominal / max. Voltage DC</th>
<th>V</th>
<th>120/148</th>
</tr>
</thead>
<tbody>
<tr>
<td>Nominal / max. output power</td>
<td>kW</td>
<td>12/25</td>
</tr>
<tr>
<td>Nominal / max. current</td>
<td>A</td>
<td>180/250</td>
</tr>
<tr>
<td>Nominal / max. torque</td>
<td>Nm</td>
<td>57/95</td>
</tr>
<tr>
<td>Nominal / max. rotation speed</td>
<td>rpm</td>
<td>2000/2500</td>
</tr>
<tr>
<td>Weight</td>
<td>kg</td>
<td>22</td>
</tr>
</tbody>
</table>

Figure 3: Cross-section of electrical motor for paraglider 1 – rotor slots for magnets; 2 – iron rotor core; 3 – stator slots; 4 – iron stator core.

The eight radially magnetized permanent magnets type NdFeB are built into the rotor core. The electric motor was put in the specially designed aluminium frame which has special air duct (Fig. 4).

3 Battery Pack and frequency converter

Electric drives for light airplane or paraglider needs the special accumulator. According to technology, the optimal solution is lithium – ion or lithium – polymer. The battery pack has a special management system (BMS) which monitors parameters of the cells like temperature and voltage of each cell, process of charging and discharging battery, and balances the cells etc. Parameters of the battery pack are presented in Table 2. The battery pack mounted in the paraglider is shown in Fig. 5.

Figure 4: The frame of electric motor mounted in paraglider

Figure 5: The battery pack mounted in paraglider under the seat of the passenger

The battery pack has special air duct, which provides cooling during operation.
Table 2: Battery Pack parameters

<table>
<thead>
<tr>
<th>Parameter</th>
<th>Unit</th>
<th>Value</th>
</tr>
</thead>
<tbody>
<tr>
<td>Capacity of the energy</td>
<td>kWh</td>
<td>7</td>
</tr>
<tr>
<td>Max voltage (full charged)</td>
<td>V</td>
<td>148</td>
</tr>
<tr>
<td>Min voltage</td>
<td>V</td>
<td>105</td>
</tr>
<tr>
<td>Max current</td>
<td>A</td>
<td>250</td>
</tr>
<tr>
<td>Charging time (up to 80% capacity)</td>
<td>hr</td>
<td>3.5</td>
</tr>
<tr>
<td>Charging time (up to 100% capacity)</td>
<td>hr</td>
<td>5</td>
</tr>
<tr>
<td>Max operating temperature</td>
<td>ºC</td>
<td>30</td>
</tr>
<tr>
<td>Weight</td>
<td>kg</td>
<td>62</td>
</tr>
</tbody>
</table>

In the electrical propulsion intended for light airplane or paraglider the special frequency converter was used. MOSFET SKAI module, manufactured by SEMIKRON, works supplied from the lithium-ion battery pack. Parameters of the frequency converter, which was mounted in paraglider under the electric motor (fig4), are presented in Table 3.

Table 3: The frequency converter parameters

<table>
<thead>
<tr>
<th>Parameter</th>
<th>Unit</th>
<th>Value</th>
</tr>
</thead>
<tbody>
<tr>
<td>SKAI 4201MD20-1450W</td>
<td></td>
<td></td>
</tr>
<tr>
<td>Max operating DC link voltage</td>
<td>V</td>
<td>160</td>
</tr>
<tr>
<td>Max inverter voltage AC</td>
<td>V</td>
<td>75</td>
</tr>
<tr>
<td>Inverter current AC</td>
<td>A</td>
<td>370</td>
</tr>
<tr>
<td>Max. inverter current AC ≤ 20sec</td>
<td>A</td>
<td>420</td>
</tr>
<tr>
<td>Max. switching frequency</td>
<td>kHz</td>
<td>20</td>
</tr>
<tr>
<td>Operating temperature</td>
<td>ºC</td>
<td>-40 to +175</td>
</tr>
<tr>
<td>Max auxiliary supply voltage DC</td>
<td>V</td>
<td>18</td>
</tr>
<tr>
<td>Weight</td>
<td>kg</td>
<td>3</td>
</tr>
</tbody>
</table>

4 Optimization process

During the optimization process influence of 2 aspects (phases) on the efficiency of the particular components, i.e. on efficiency of the motor and frequency converter separately, as well as on efficiency of the whole drive (motor + converter) were investigated: First phase - switching frequency of the converter - adjusting of the switching frequency changes switching losses in the converter and the additional losses in the motor. Second phase – propellers - 3 different propellers (1 fixed, 2 adjustable) were used and tested. The examinations process of an influence of switching frequency on the efficiency, two switching frequencies – 10 kHz and 15 kHz were set in the converter. According to literature, switching frequency influences on working time of MOSFET transistors what causes changing commutation losses of the transistors, what was proven. Additional losses of electric motor rise, but the efficiency of the entire propulsion practically do not change (fig8). By changing the switching frequency only the distribution of losses can be changed. Differences between efficiency of inverter and motor (separately) were presented in figure 6 and 7.

![Figure 6: Comparison of η=f(T) – only inverter](image)

![Figure 7: Comparison of η=f(T) – only electric motor](image)

![Figure 8: Comparison of η=f(T) – inverter + motor](image)
15 kHz switching frequency. From the characteristic comparison of efficiency vs. torque \{fig8\}, it can be seen the efficiency of complete drive system (inverter+motor) practically do not change.

![Figure 9: Comparison of η=f(T) – inverter + motor for 10 kHz switching frequency](image)

![Figure 10: Comparison of η=f(T) – inverter + motor for 15 kHz switching frequency](image)

Efficiency map for 10 kHz switching frequency (inverter+motor) is shown in figure 9. The highest efficiency of the propulsion system is obtained at the range of rated torque between 20 to 60 Nm and rotational speed between 1900 to 2400 rpm. The very similar situation, like in case of figure 9, is for 15 kHz switching frequency for inverter and motor \{fig10\}. The area with highest efficiency appears for 20 to 60 Nm and rotational speed between 1800 to 2500 rpm. After performing the first flight, which provided many data needed for further research, an influence of different propellers (1 fixed, 2 adjustable) was investigated. Results of tests are presented in figure 11.

![Figure 11: Comparison of Thrust/Power=f(n) characteristics](image)

The fixed Helix propeller has larger thrust than remaining examined propellers but it is also louder. The fixed Helix propeller has two blades and it is not adjustable. The two remaining investigated propellers have three adjustable blades. The angle of three-bladed propellers was set to reach thrust of 100 KG at maximal rotation speed. The electric paraglider was tested in air during two persons flight. The course of output power taken from battery during the flight was presented in figure 12.

![Figure 12: Output power taken from battery during the two persons flight](image)

During the two persons flight which had about 25 minutes, 2.6 kWh energy were used. The weight of pilot and passenger was 90 and 60 KG.

5 Conclusion

As can be recognized from figure 7 the electric motor at the 15 kHz switching frequency has the efficiency higher than 90% in range of torque operation between 15 to 70 Nm. In case of electric motor controlled with 10 kHz switching frequency, situation is very similar. The efficiency is above 90%. The difference of efficiency in whole range operation torque for converter \{fig.9\} is more visible. The highest efficiency of the inverter is for 10 kHz switching frequency, but from figure 8 it
can be noticed that there is no differences between characteristics of the efficiency vs. load torque for 10 and 15 kHz switching frequency if the whole propulsion system is investigated. From examinations done and received results (Fig. 9 and 10, efficiency maps), it can be seen, the area with highest efficiency is practically the same for two examined cases (10 and 15 kHz switching frequency). The range of torque and rotational speed (for highest efficiency) do not change, even for different switching frequency. After adjusting switching frequency, only the distribution of losses can be changed. The second phase of investigations proved that the propellers have higher impact factor into efficiency level of the drive than the switching frequency. The fixed Helix propeller has a better coefficient (Thrust/Power) than two remaining examined propellers (Fig. 11). The Fixed Helix propeller produce higher thrust (in range 1800 to 2400 rpm) with smaller power demand than Peszke and Helix adjustable three-bladed propellers. Producing the higher thrust by Fixed Helix propeller (in range 1800 to 2400 rpm - Fig. 10) causes higher noise level. The whole electrical propulsion system mounted in paraglider was tested in air as well. It enables to fly environment friendly with no pollution and no danger of flammable fuel. The noise of the system is lower than the combustion engine drive, what is very important as well.

References


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